

Example

A notched aluminum component consists of a plate 12 in. wide and 1 in. thick with a 2 in. diameter center hole. Determine the crack initiation life of the component under the axial stress histories shown below. History A is a constant amplitude zero-to-maximum ($R=0$) axial stress. Histories B and C have initial overloads followed by the same constant amplitude zero-to-maximum stress fluctuations.

Geometry: $W = 12$ in. $t = 1$ in. $d = 2$.in, $K_t = 2.56$

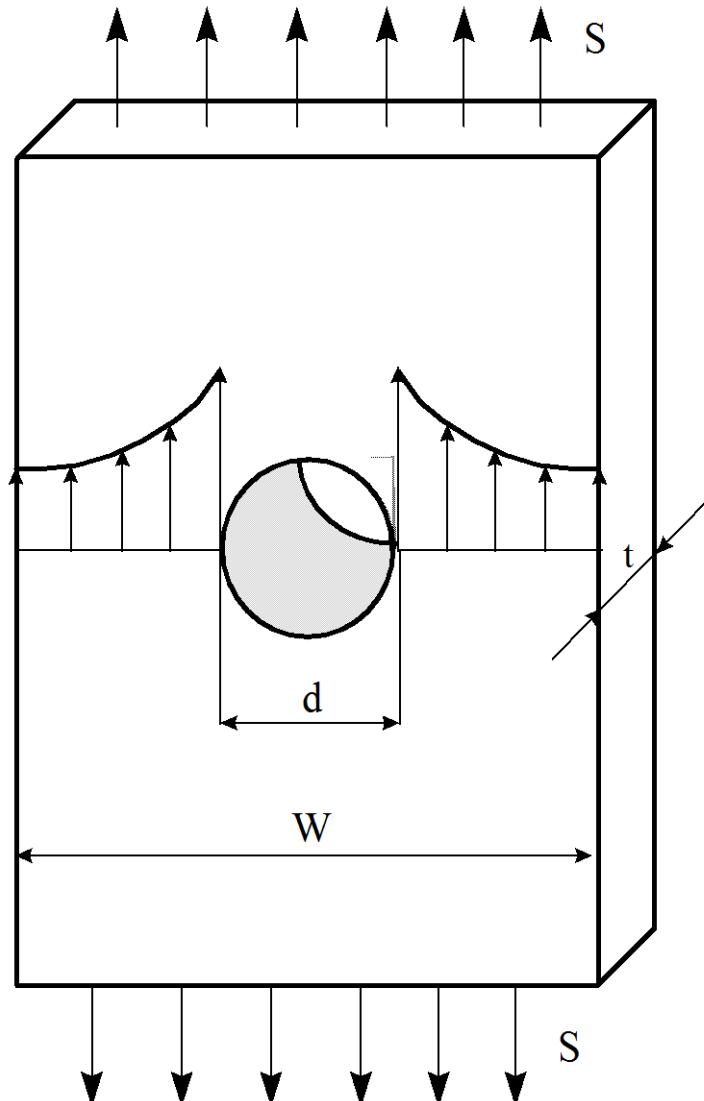
Material;

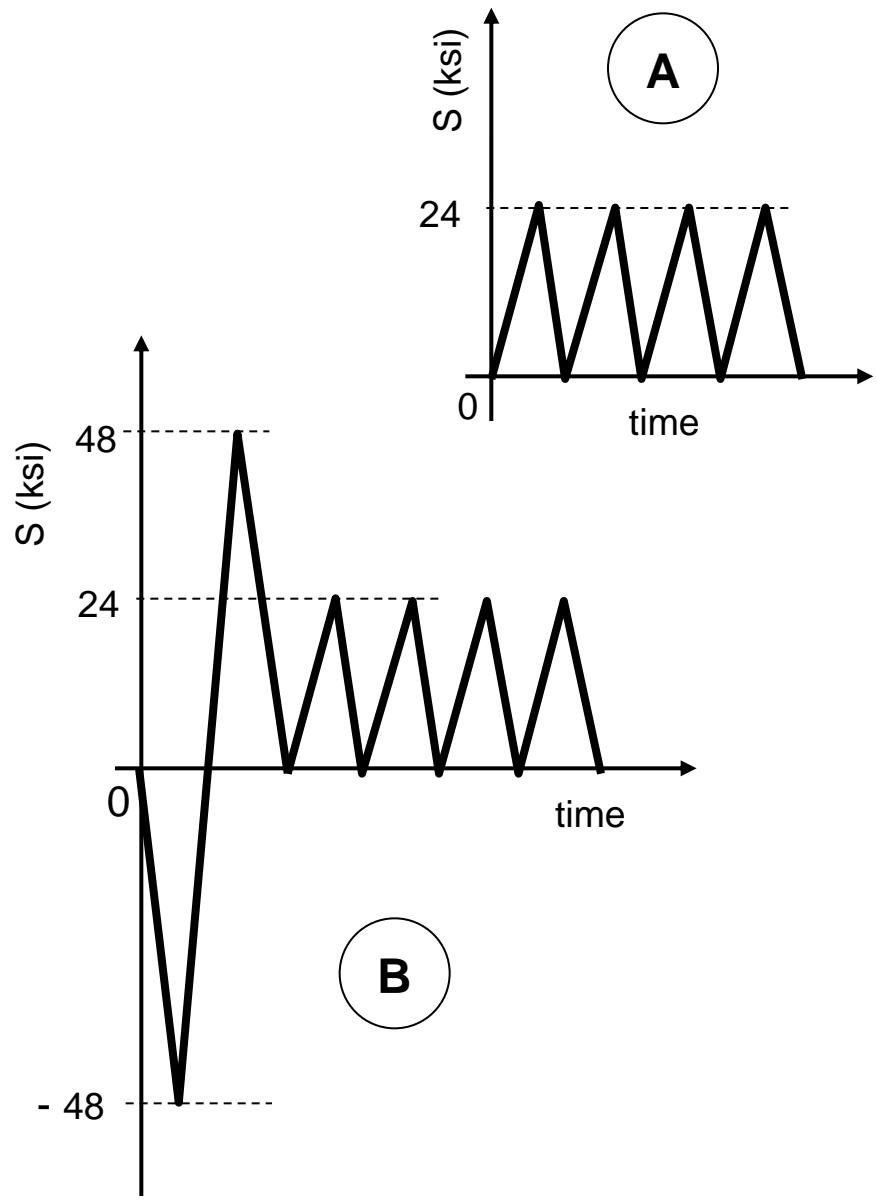
Cyclic stress – strain curve:

$E = 10600$ ksi, $K = 95$ ksi, $n = 0.065$

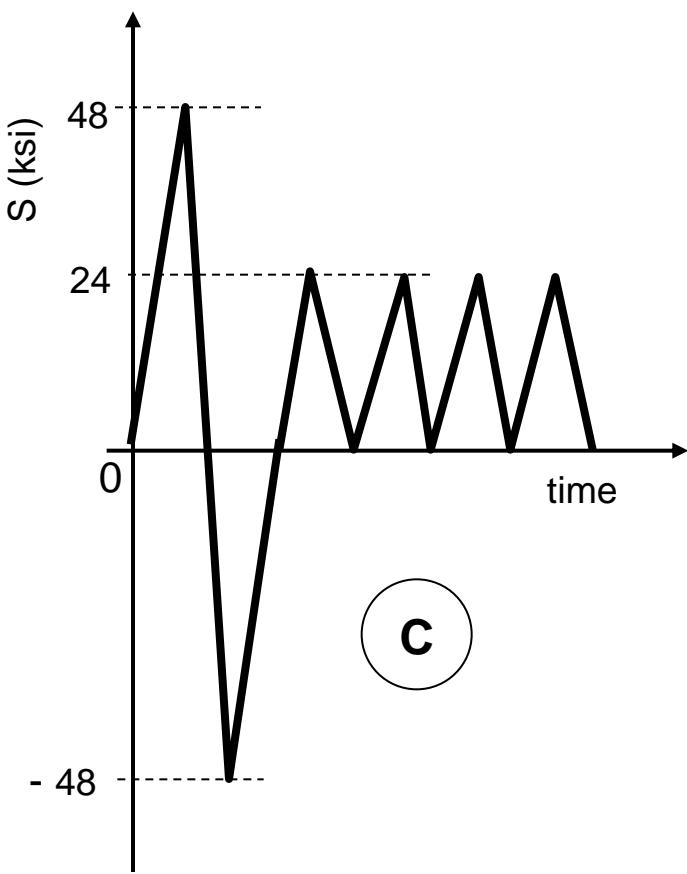
Fatigue strain-life curve:

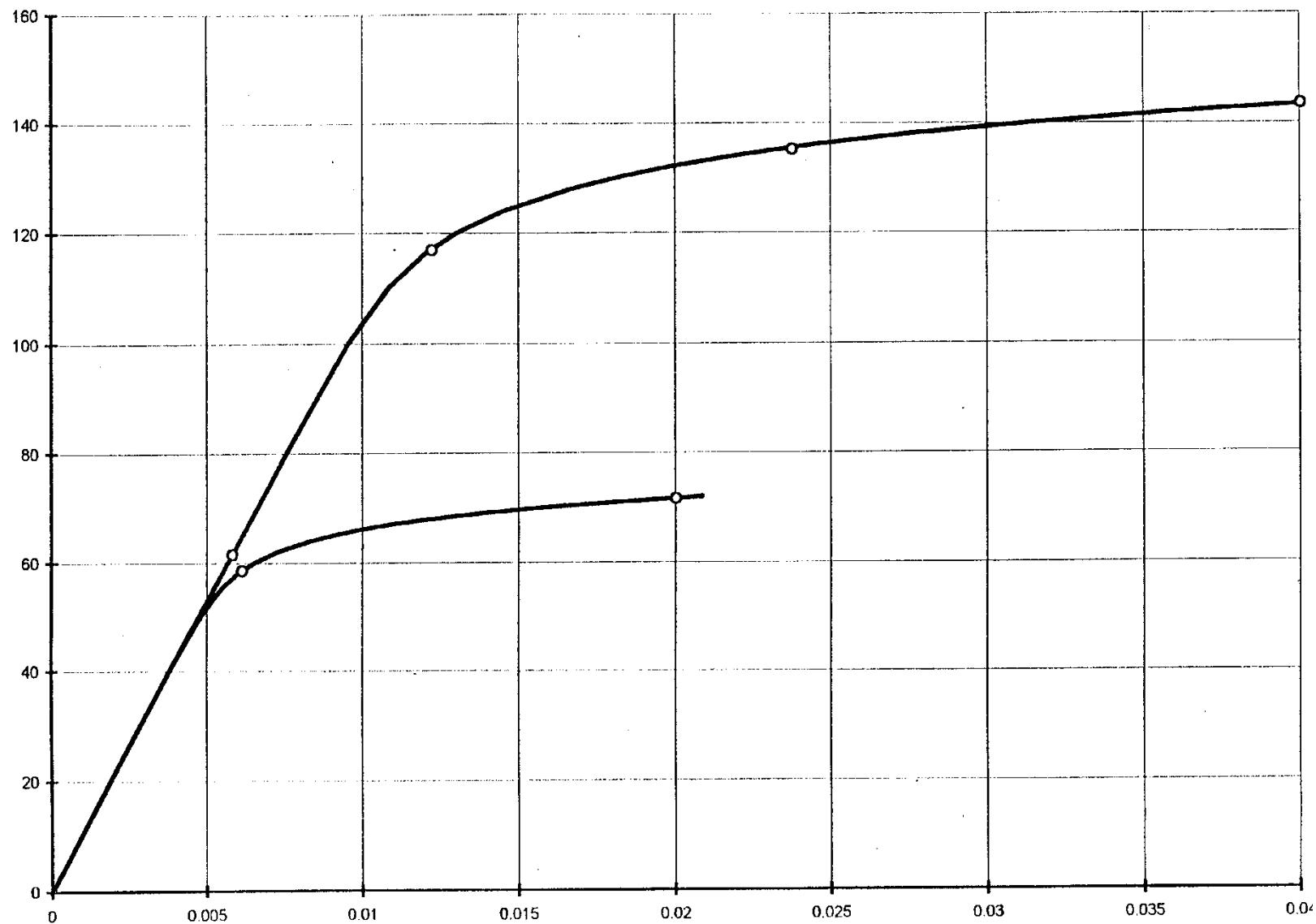
$\sigma_f = 160$ ksi, $b = -0.124$, $\varepsilon_f = 0.22$, $c = -0.59$

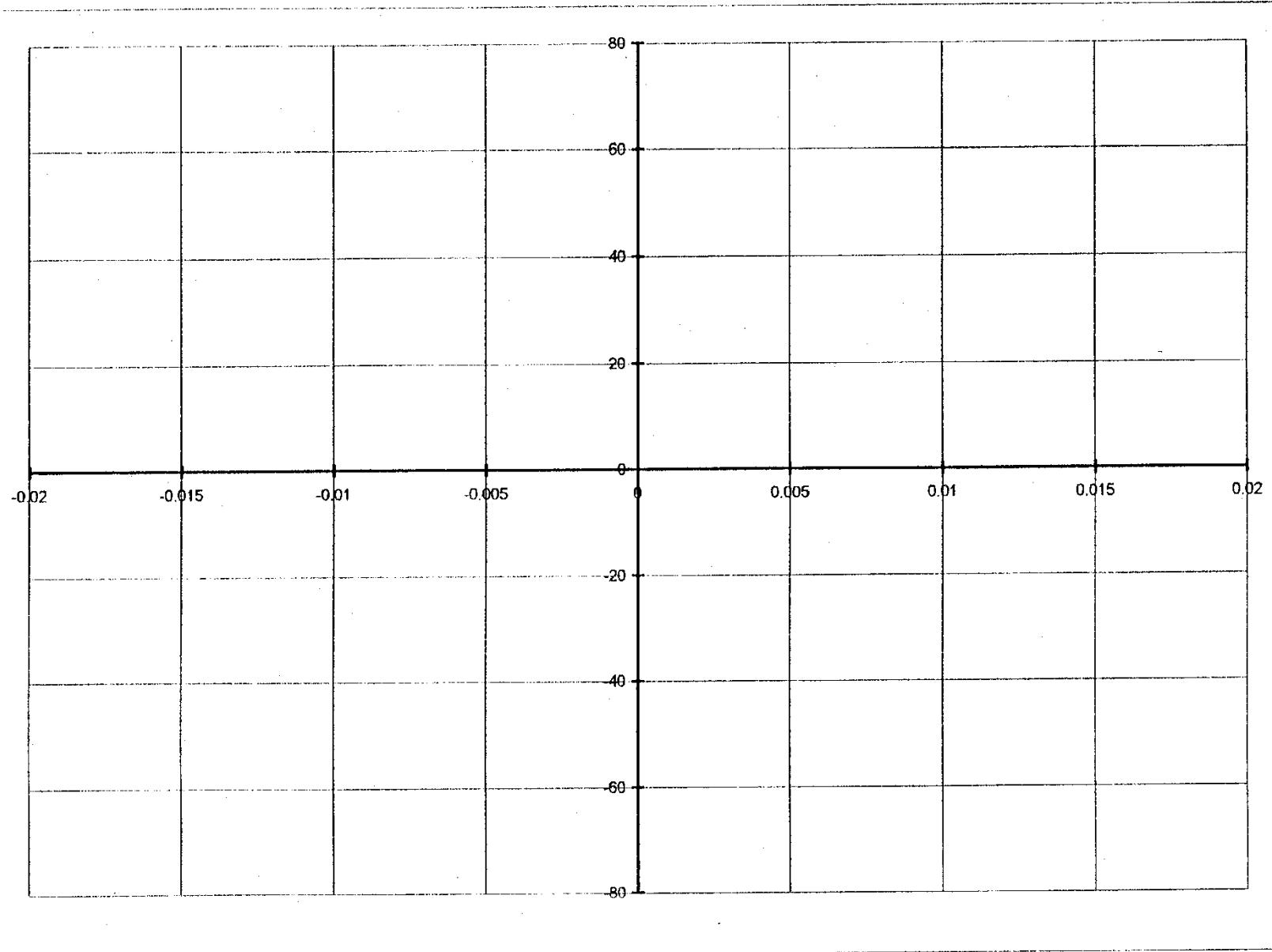


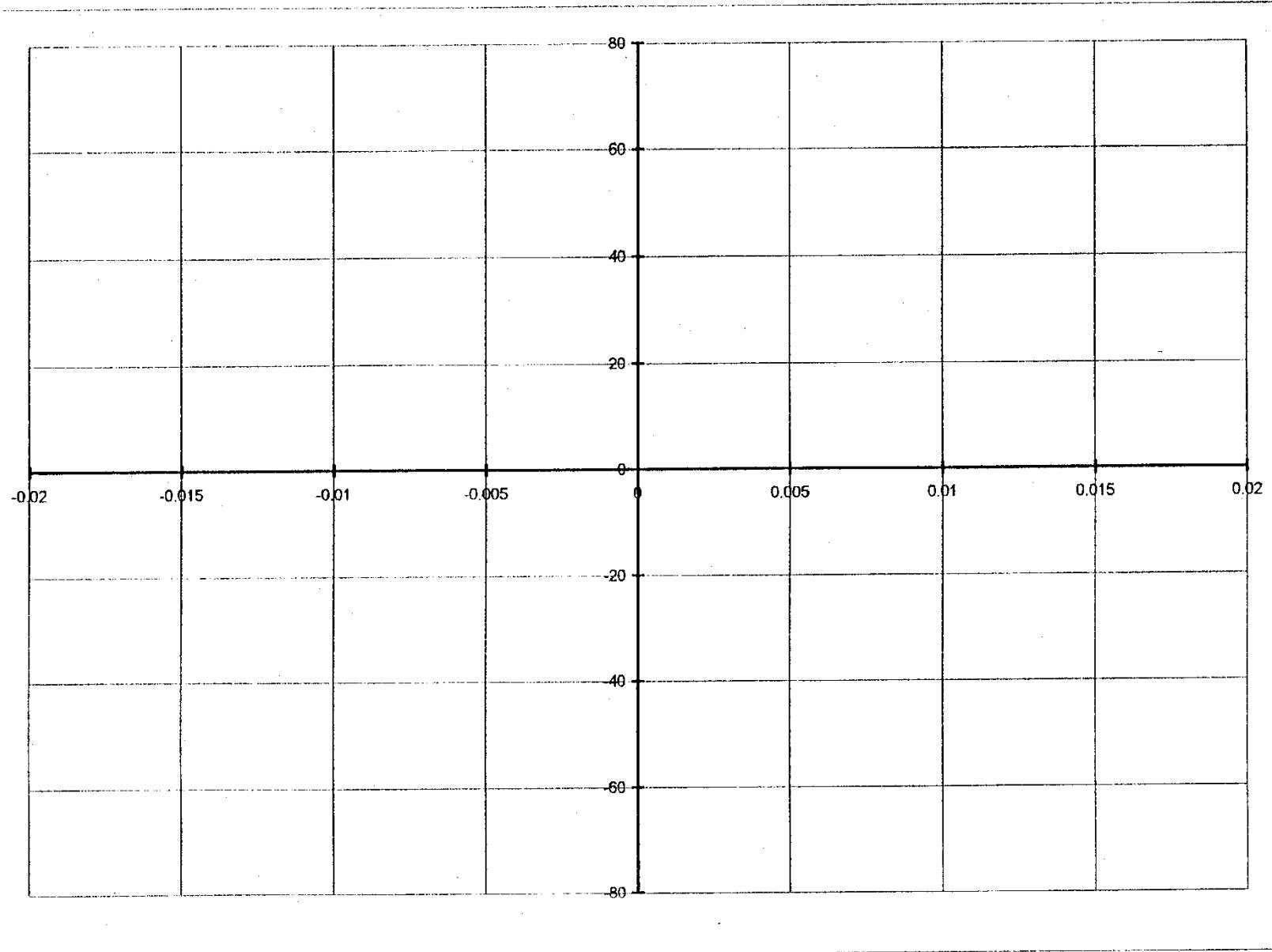


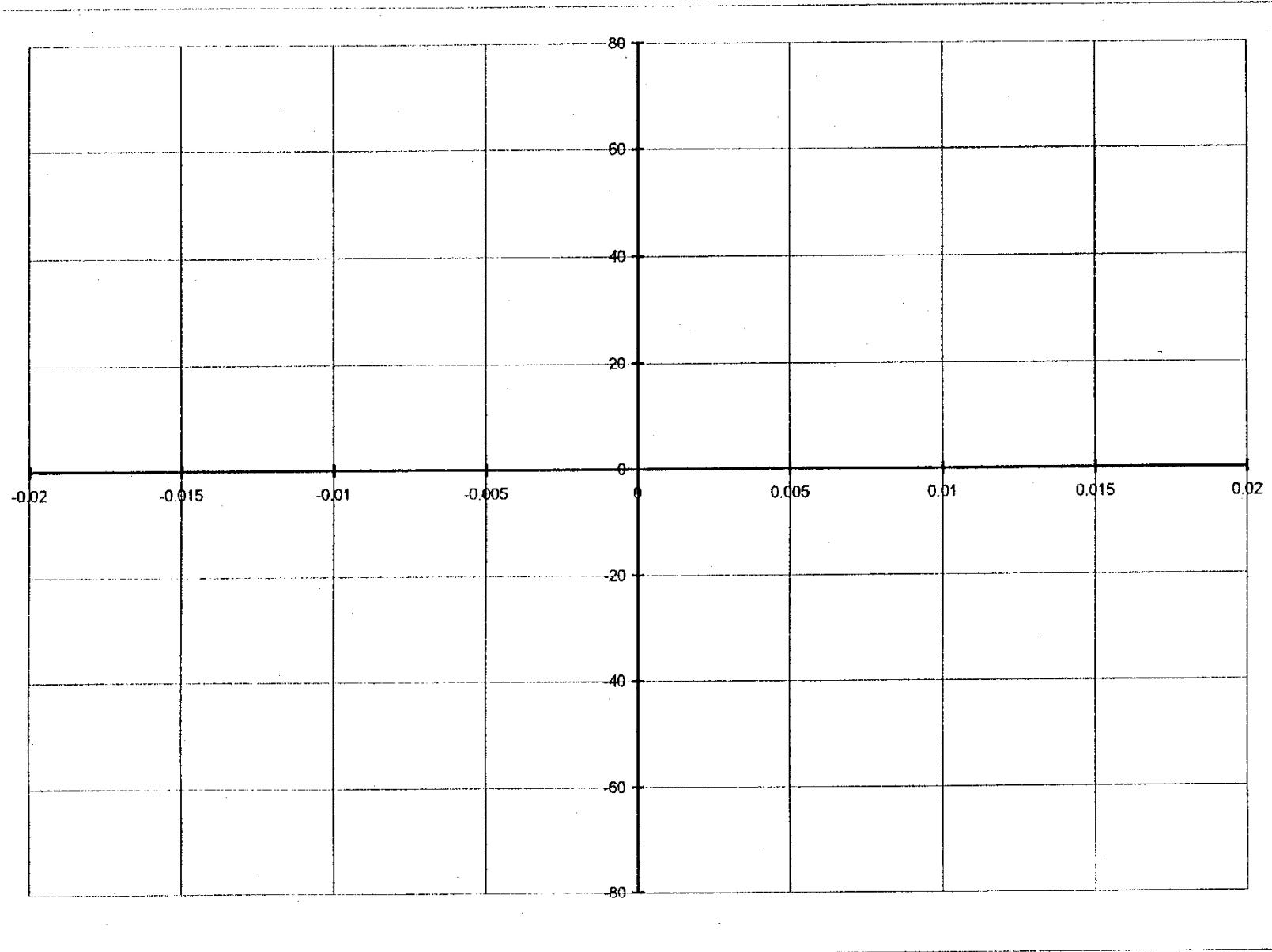
Nominal stress histories











Stress history A

1st reversal 0-24 ksi (the Neuber rule)

$$\begin{cases} \frac{(K_t S_{\max})^2}{E} = \sigma_{\max} \epsilon_{\max} \\ \epsilon_{\max} = \frac{\sigma_{\max}}{E} + \left(\frac{\sigma_{\max}}{K} \right)^{\frac{1}{n}} \end{cases} \Rightarrow \begin{cases} \frac{(2.56 \times 24)^2}{10600} = \sigma_{\max} \epsilon_{\max} \\ \epsilon_{\max} = \frac{\sigma_{\max}}{10600} + \left(\frac{\sigma_{\max}}{95} \right)^{\frac{1}{0.065}} \end{cases} \Rightarrow \begin{cases} \sigma_{\max} = 59.36 \text{ ksi} \\ \epsilon_{\max} = 0.0060022 \end{cases}$$

2nd reversal 24-0 ksi

$$\Delta S = |S_{\max} - S_{\min}| = |24 - 0| = 24 \text{ ksi}$$

$$\begin{cases} \frac{(K_t \Delta S)^2}{E} = \Delta \sigma \cdot \Delta \epsilon \\ \frac{\Delta \epsilon}{2} = \frac{\Delta \sigma}{2E} + \left(\frac{\Delta \sigma}{2K} \right)^{\frac{1}{n}} \end{cases} \Rightarrow \begin{cases} \frac{(2.56 \times 24)^2}{10600} = \Delta \sigma \cdot \Delta \epsilon \\ \frac{\Delta \epsilon}{2} = \frac{\Delta \sigma}{2 \cdot 10600} + \left(\frac{\Delta \sigma}{2 \cdot 95} \right)^{\frac{1}{0.065}} \end{cases} \Rightarrow \begin{cases} \Delta \sigma = 61.45 \text{ ksi} \\ \Delta \epsilon = 0.0057979 \end{cases}$$

Minimum stress and strain after the 2nd reversal

$$\begin{cases} \sigma_{\min} = \sigma_{\max} - \Delta\sigma \\ \varepsilon_{\min} = \varepsilon_{\max} - \Delta\varepsilon \end{cases} \Rightarrow \begin{cases} \sigma_{\min} = 59.36 - 61.45 \\ \varepsilon_{\min} = 0.006022 - 0.0057979 \end{cases} \Rightarrow \begin{cases} \sigma_{\min} = -2.09 \text{ ksi} \\ \varepsilon_{\min} = 0.0002043 \end{cases}$$

3rd reversal 0-24 ksi

$$\Delta S = |S_{\max} - S_{\min}| = |0 - 24| = 24 \text{ ksi}$$

$$\begin{cases} \frac{(K_t \Delta S)^2}{E} = \Delta\sigma \cdot \Delta\varepsilon \\ \frac{\Delta\varepsilon}{2} = \frac{\Delta\sigma}{2E} + \left(\frac{\Delta\sigma}{2K}\right)^{\frac{1}{n}} \end{cases} \Rightarrow \begin{cases} \frac{(2.56 \times 24)^2}{10600} = \Delta\sigma \cdot \Delta\varepsilon \\ \frac{\Delta\varepsilon}{2} = \frac{\Delta\sigma}{2 \cdot 10600} + \left(\frac{\Delta\sigma}{2 \cdot 95}\right)^{\frac{1}{0.065}} \end{cases} \Rightarrow \begin{cases} \Delta\sigma = 61.45 \text{ ksi} \\ \Delta\varepsilon = 0.0057979 \end{cases}$$

Maximum stress and strain after the 3rd reversal

$$\begin{cases} \sigma_{\max} = \sigma_{\min} + \Delta\sigma \\ \varepsilon_{\max} = \varepsilon_{\min} + \Delta\varepsilon \end{cases} \Rightarrow \begin{cases} \sigma_{\max} = -2.09 + 61.45 \\ \varepsilon_{\max} = 0.0002043 + 0.0057979 \end{cases} \Rightarrow \begin{cases} \sigma_{\max} = 59.36 \text{ ksi} \\ \varepsilon_{\max} = 0.006022 \end{cases}$$

The Maximum stress and strain after the 1st and the 3rd reversal are the same.
 The stress –strain loop has been closed, i.e. the stress strain cycle has been found!

$$\begin{cases} \sigma_{\max} = 59.36 \text{ ksi} \\ \varepsilon_{\max} = 0.006022 \end{cases} \quad \begin{cases} \sigma_{\min} = -2.09 \text{ ksi} \\ \varepsilon_{\min} = 0.0002043 \end{cases} \quad \begin{cases} \Delta\sigma = 61.45 \text{ ksi} \\ \Delta\varepsilon = 0.0057979 \end{cases}$$

$$\sigma_{mean} = \frac{\sigma_{\max} + \sigma_{\min}}{2} = \frac{59.36 - 2.09}{2} = 28.635 \text{ ksi}$$

The Manson-Coffin curve corrected for the mean stress effect (Morrow's correction)

$$\frac{\Delta\varepsilon}{2} = \frac{\sigma' - \sigma_{mean}}{E} (2N_f)^b + \varepsilon'_f (2N_f)^c$$

$$\frac{0.0057979}{2} = \frac{160 - 28.635}{10600} (2N_f)^{-0.124} + 0.22 (2N_f)^{-0.59}$$

↓

$N_f = 97880$ cycles; Calculated

$N_f = 110000$ cycles; Experimental by NASA